### Information-Aware Powered Descent Guidance for Entry, Descent, and Landing Christopher R. Hayner<sup>1</sup>, Natalia Pavlasek<sup>1</sup>,

Karen Leung<sup>1</sup>, Behçet Açıkmeşe<sup>1</sup>, John M. Carson III<sup>2</sup>

<sup>1</sup>William E. Boeing Department of Aeronautics & Astronautics, University of Washington, Seattle WA <sup>2</sup>NASA Johnson Space Center, Houston TX







Task: Given no *a priori* information about the terrain, we seek to build an model of the terrain to inform PDG algorithms



#### **Exploration:**

Explore the environment to obtain information to facilitate a safe landing that satisfies mission objectives

#### **Exploration:**

Explore the environment to obtain information to facilitate a safe landing that satisfies mission objectives

**Exploitation:** Exploit information to determine a safe landing site and commit to landing

## Exploration: Explore the environment to obtain information to factor a safe landing that sufficients mission objectives

Exploitation: Work on exploitation phase Tomitatet al. 2024, Buckner et al, 2024, Hayner et al, 2023 and commit to landing Exploration: Explore the environment to obtain information to facilitate a safe landing that satisfies mission objectives



Exploration: Explore the environment to obtain information to facilitate a safe landing that satisfies mission objectives



Exploration: Explore the environment to obtain information to facilitate a safe landing that satisfies mission objectives



#### Map Representation.

#### The map is represented using Gaussian process regression.

Given samples of a function, **Gaussian process regression** finds a distribution of functions that the samples could have been drawn from. A Gaussian process is defined by a mean function and a covariance function.



Information map is obtained by fitting a Gaussian Process (GP) to point cloud data



Information map is obtained by fitting a Gaussian Process (GP) to point cloud data

- Covariance of GP over the full map











### Greedy





#### Map Discretization.

The Gaussian process is sampled over a fine grid. The covariance of the points is binned and averaged to get a coarse grid.



Finely sampled Gaussian process.



Binned map.

#### Map Discretization.

The Gaussian process is sampled over a fine grid. The covariance of the points is binned and averaged to get a coarse grid.



Finely sampled Gaussian process.



Binned map.

To model map covariance, we make the following assumptions:

To model map covariance, we make the following assumptions:

#### **Assumption 1:**

When a given region of the environment is observed, its covariance will decrease and, when it is not observed, it will remain constant.

To model map covariance, we make the following assumptions:

#### **Assumption 1:**

When a given region of the environment is observed, its covariance will decrease and, when it is not observed, it will remain constant.

#### **Assumption 2:**

The map covariance is assumed to be uniformly decreased across the sensor's frame of view (FoV).

First, we define the norm cone  $\,\mathcal{K}\,$ 

 $||A^C p_{\mathcal{S}}|| \le c^\top p_{\mathcal{S}}$ Convex in  $\mathcal{P}_{\mathcal{S}}$ 

First, we define the norm cone  $\,\mathcal{K}\,$ 

$$||A^C p_{\mathcal{S}}|| \leq c^\top p_{\mathcal{S}}$$
 Convex in  $p_{\mathcal{S}}$ 



















Lastly, to ensure that the covariance doesn't become negative, we use the following piecewise function.

$$\dot{\sigma}(p_{\mathcal{S}},t) = \begin{cases} \left(\frac{1-\delta}{1+e^{-\alpha \max\{0,\mathcal{K}(p_{\mathcal{S}})\}^2}} - (1-\delta)\right) & \text{if } \sigma(p_{\mathcal{S}},t) \ge 0\\ 0 & \text{otherwise} \end{cases}$$

#### System Dynamics

We use a 6-Degree of Freedom rigid body model with mass dynamics.

Mass.	$\dot{m} = -\alpha   T_{\mathcal{B}}(t)  _2,$
Position.	$\dot{r}_{\mathcal{I}}(t) = v_{\mathcal{I}}(t),$
Velocity.	$\dot{v}_{\mathcal{I}}(t) = \frac{1}{m} \left( C_{\mathcal{B} \to \mathcal{I}}(q_{\mathcal{B} \to \mathcal{I}}(t)) T_{\mathcal{B}}(t) \right) + g_{\mathcal{I}},$
Attitude.	$\dot{q}_{\mathcal{I}\to\mathcal{B}} = \frac{1}{2}\Omega(\omega_{\mathcal{B}}(t))q_{\mathcal{I}\to\mathcal{B}}(t),$
Angular Velocity.	$\dot{\omega}_{\mathcal{B}}(t) = J_{\mathcal{B}}^{-1} \left( M_{\mathcal{B}}(t) - \left[ \omega_{\mathcal{B}}(t) \times \right] J_{\mathcal{B}} \omega_{\mathcal{B}}(t) \right)$

#### System Dynamics

We use a 6-Degree of Freedom rigid body model with mass dynamics.

$$\begin{split} & \text{Mass.} & \dot{m} = -\alpha ||T_{\mathcal{B}}(t)||_{2}, \\ & \text{Position.} & \dot{r}_{\mathcal{I}}(t) = v_{\mathcal{I}}(t), \\ & \text{Velocity.} & \dot{v}_{\mathcal{I}}(t) = \frac{1}{m} \left( C_{\mathcal{B} \to \mathcal{I}}(q_{\mathcal{B} \to \mathcal{I}}(t)) T_{\mathcal{B}}(t) \right) + g_{\mathcal{I}}, \\ & \text{Attitude.} & \dot{q}_{\mathcal{I} \to \mathcal{B}} = \frac{1}{2} \Omega(\omega_{\mathcal{B}}(t)) q_{\mathcal{I} \to \mathcal{B}}(t), \\ & \text{Angular Velocity.} & \dot{\omega}_{\mathcal{B}}(t) = J_{\mathcal{B}}^{-1} \left( M_{\mathcal{B}}(t) - [\omega_{\mathcal{B}}(t) \times] J_{\mathcal{B}} \omega_{\mathcal{B}}(t) \right), \\ & \text{Map Covariance.} & \dot{\sigma}(p_{\mathcal{S}}, t) = \begin{cases} \left( \frac{1-\delta}{1+e^{-\alpha \max\{0, \mathcal{K}(p_{\mathcal{S}})\}^{2}} - (1-\delta) \right) & \text{if } \sigma(p_{\mathcal{S}}, t) \geq 0 \\ 0 & \text{otherwise} \end{cases}, \end{split}$$

#### Nonlinear Trajectory Optimization Problem.

$$\min_{x,u} \int_{t_i}^{t_f} \sum_{i=1}^{N_{\text{grid}}} \sigma(p_{\mathcal{S},i}, t) dt$$
  
s.t.  $\dot{x}(t) = f(x(t), u(t)),$   
 $x_{\min} \leq x(t) \leq x_{\max},$   
 $u_{\min} \leq u(t) \leq u_{\max},$   
 $x(t) \in \mathcal{X},$   
 $u(t) \in \mathcal{U},$   
 $x(t_i) \in \mathcal{X}_{t_i},$ 

 $\forall t \in [t_i, t_f],$   $\forall t \in [t_i, t_f],$  $\forall t \in [t_i, t_f],$ 





#### Map Covariance



#### Key Takeaways.

- 1. We propose an algorithm for maximizing information in a PDG scenario.
- 2. We embed this covariance as a state in our dynamics and minimize the integral.
- 3. We demonstrate the effectiveness of the proposed method in a high-fidelity simulated environment.

#### Acknowledgments

- Dr. Davis Adams for his gracious support during our testing at the NASA Simulation, Emulation, Navigation, Sensing, and STAR laboratory
- NASA Space Technology Graduate Research Opportunities
- Office of Naval Research under grant N00014-17-1-2433
- Natural Sciences and Engineering Research Council of Canada (NSERC).

